

INERTIAL-VISION NAVIGATION WITH SUPPORT FROM FLAT TERRAIN MAP

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INTRODUCTION

- A method to estimate position and heading of the aerial vehicle
- Focus on the concept, not many hard results (mainly due to lack of data)
- Short introduction on Particle Filter
- Short introduction on Chamfer matching
- Evaluation of the likelihood function

MAIN IDEA

- Use downward pointing camera images together with a flat map of the environment (represented as an image) to estimate the position and heading of the flying platform
- Flat map assumes that pitch and roll attitude of the vehicle is small – helicopter or blimp-like platform
 - Even with 3D map, the ambiguity between position and attitude is still present for this configuration
- Images and map are utilised in the likelihood function
- Estimation is performed with the Particle Filter
 - Handles likelihood function in a nice way (unlike EKF)



ESTIMATION PROBLEM

- Given a system, \mathcal{S} , and some observations (measurements) created from it, \mathbb{Y} , estimate the hidden variables the are contained in the system, \mathbb{X} , with an estimator \mathcal{E} . The observations may include inputs to the system, \mathbb{U} , as well.



- Navigation is an example of the estimation problem.
- Estimation can be solved in many different ways
 - Filtering (real-time)
 - Smoothing (batch-oriented)

PARTICLE FILTER

Particle Filter (PF) solves an optimal (non-linear) filtering problem

$$p(x_t|y_{1:t}) = \frac{p(y_t|x_t)p(x_t|y_{1:t-1})}{p(y_t|y_{1:t-1})}$$
$$p(x_{t+1}|y_{1:t}) = \int p(x_{t+1}|x_t)p(x_t|y_{1:t}) dx_t$$

with a sampling method (Monte Carlo).

In the equations above:

$p(y_t|x_t)$ is the likelihood function (usually a measurement equation),

$p(x_{t+1}|x_t)$ is the state transition pdf (usually state dynamics),

$p(x_t|y_{1:t})$ is the posterior pdf (what we want to estimate),

$p(x_t|y_{1:t-1})$ is the prior pdf (prediction of the state at time t before measurements arrive).

PARTICLE FILTER

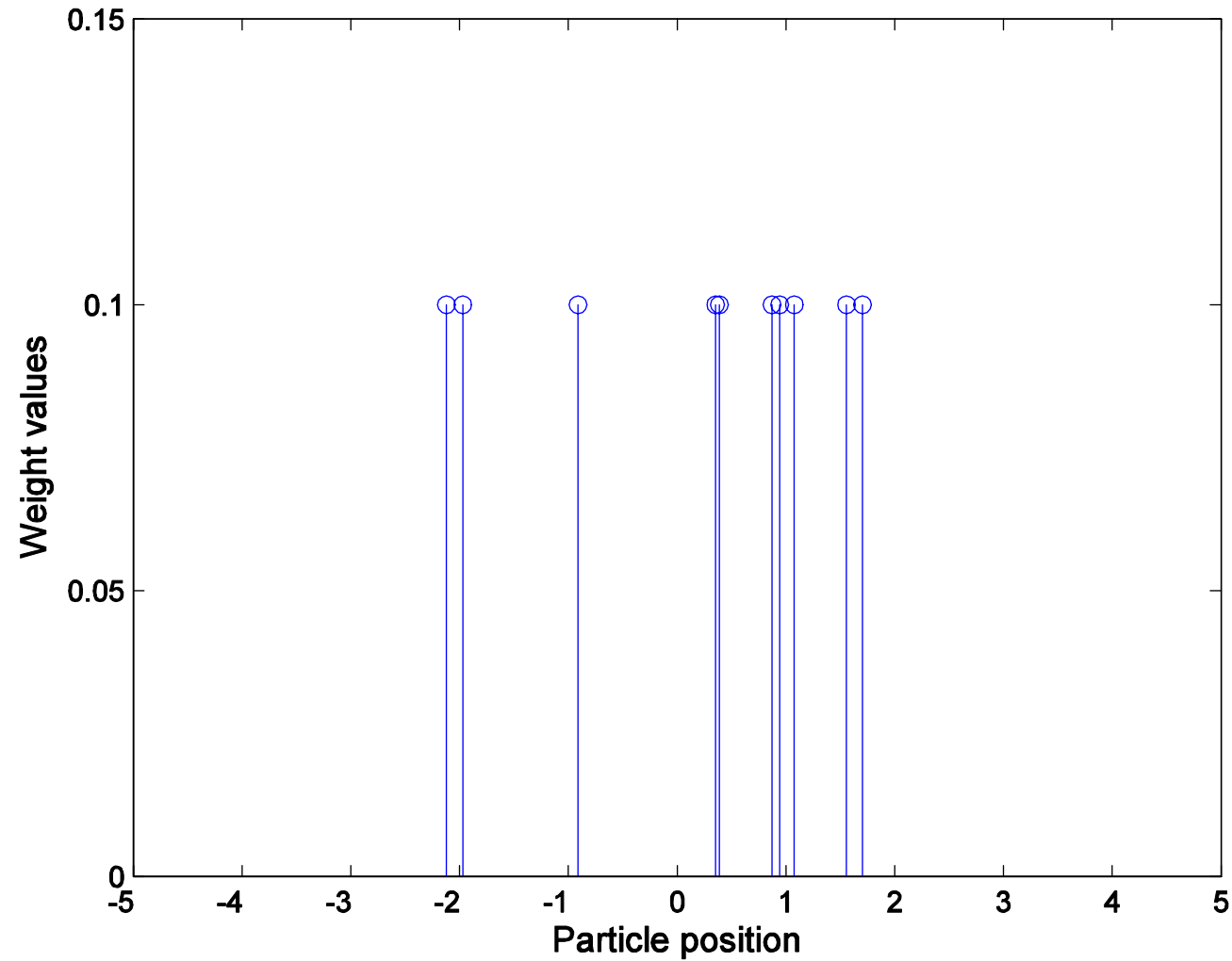
- pdfs from previous slide are approximated with N particles with corresponding weights: $\{x_t^i, w_t^i\}_{i=1:N}$
- Two main steps are iterated
 - Weight update – "measurement update"
 - Resampling and particle transition – "time update"

- Point estimate can be obtained as a weighted particle sum

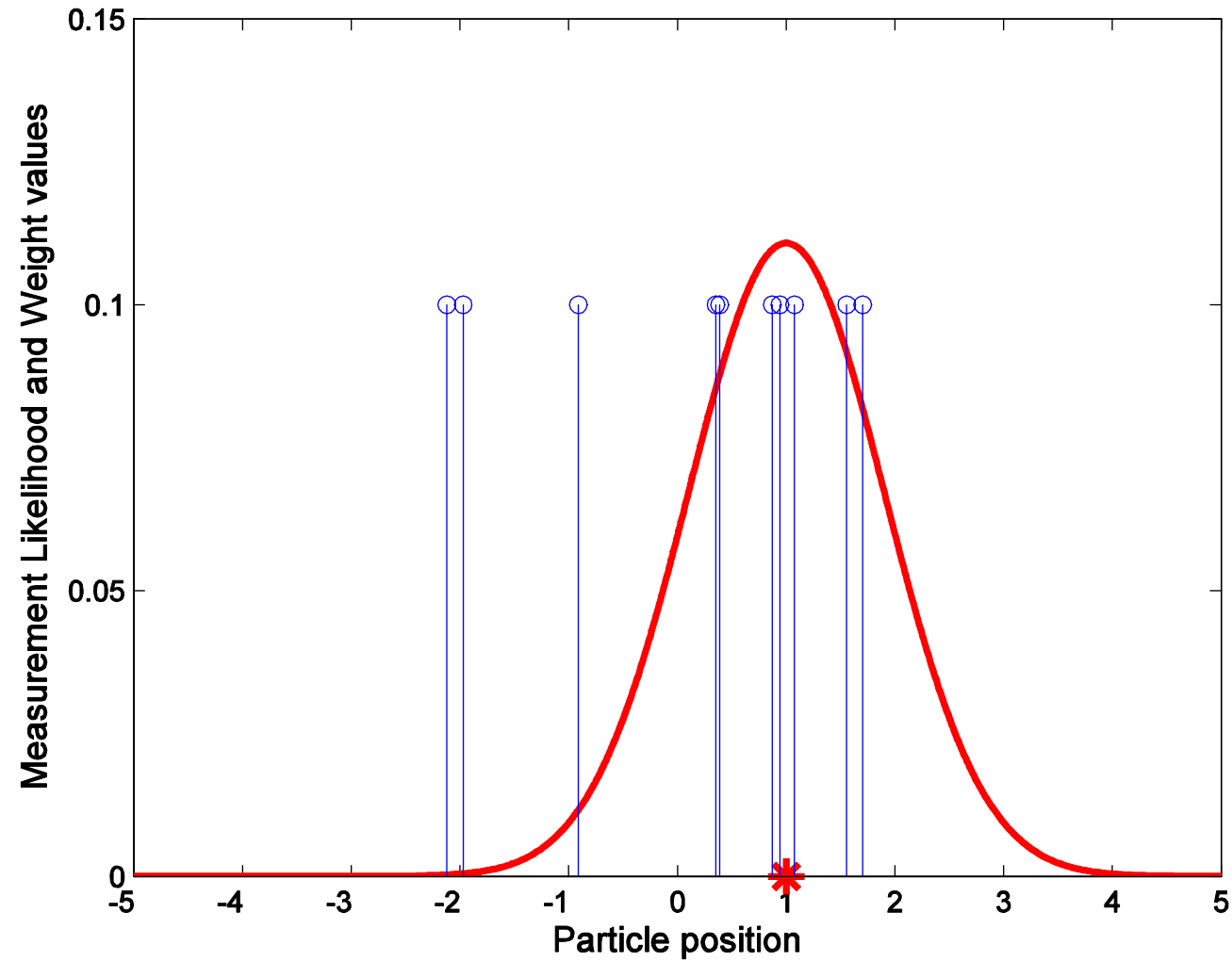
$$\hat{x}_t = \sum_{i=1}^N w_t^i x_t^i$$

- Illustration with a toy example
 - One-dimensional
 - 10 particles
 - Simple transition pdf (state dynamics):
 $p(x_{t+1}|x_t) = \mathcal{N}(x_{t+1}; x_t + u_t, Q), (x_{t+1} = x_t + u_t + w_t)$
 - Simple likelihood (measurement equation):
 $p(y_t|x_t) = \mathcal{N}(y_t; x_t, R), (y_t = x_t + e_t)$

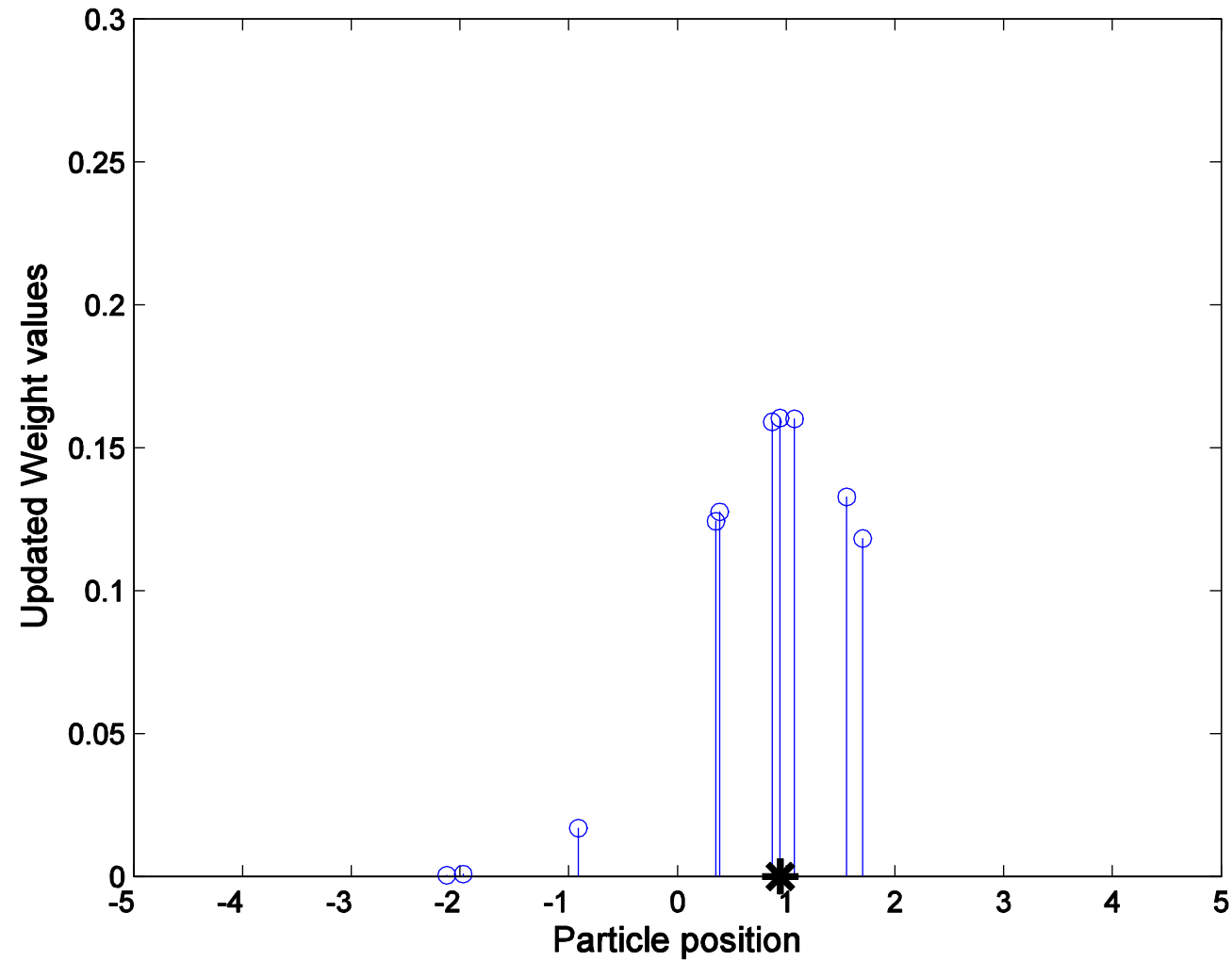
PARTICLE FILTER



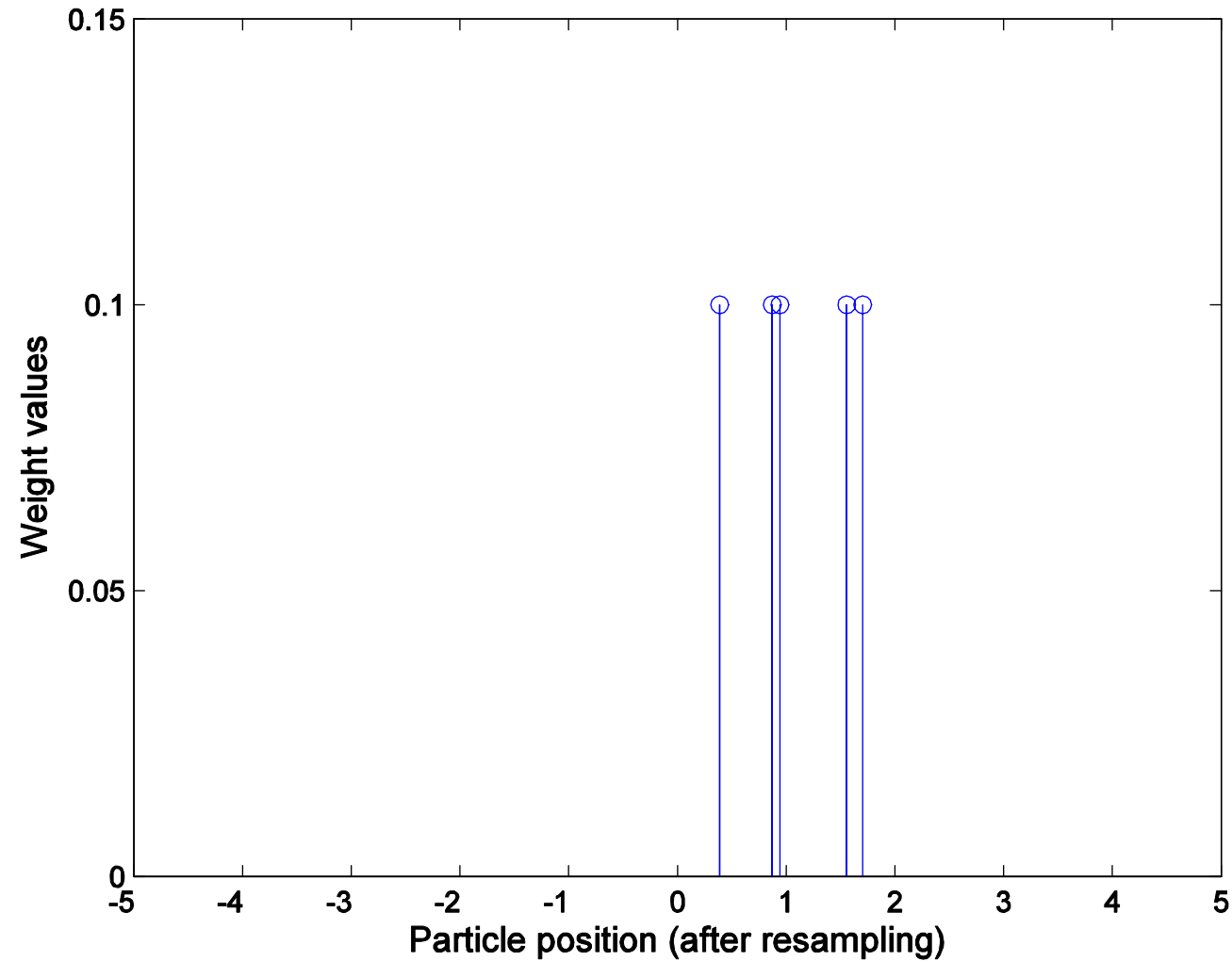
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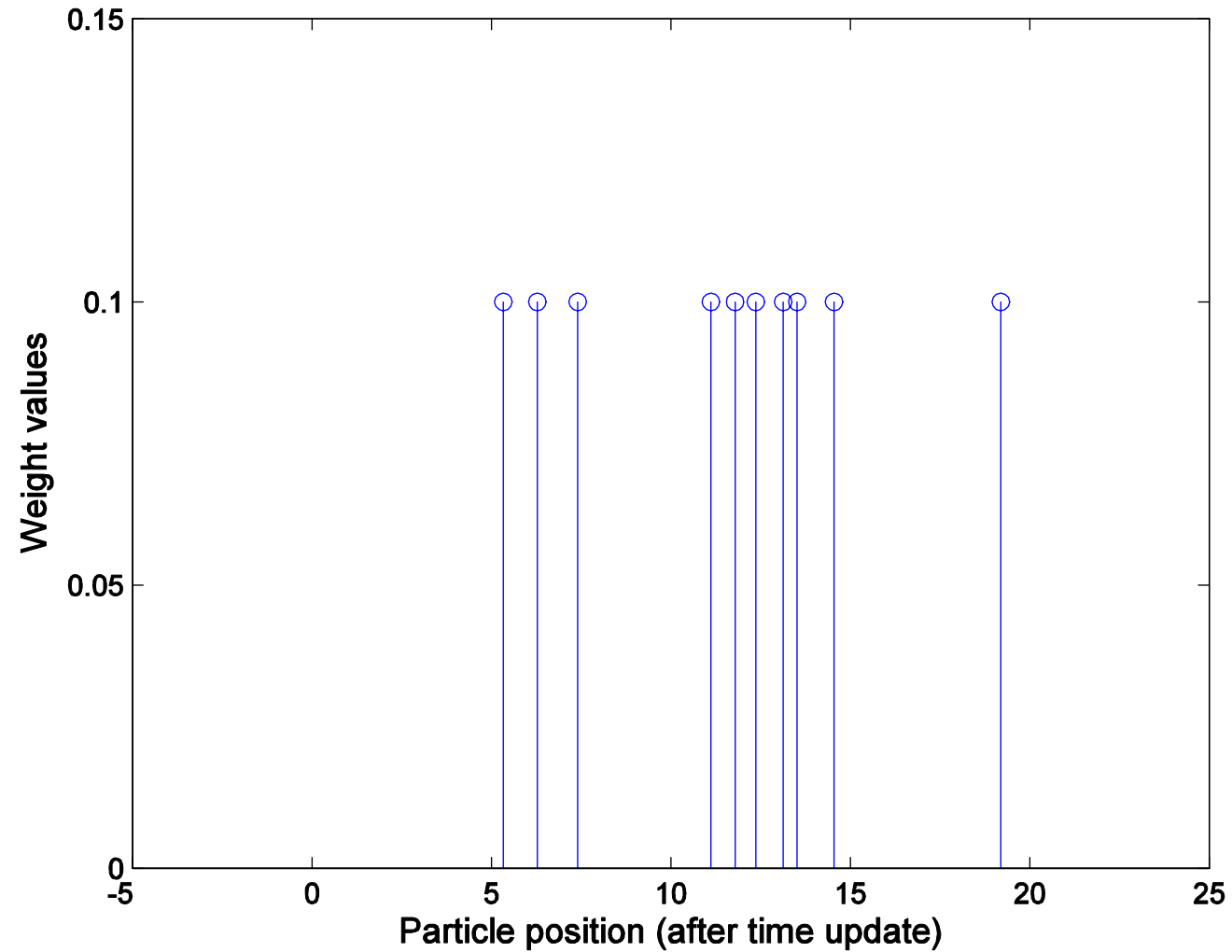
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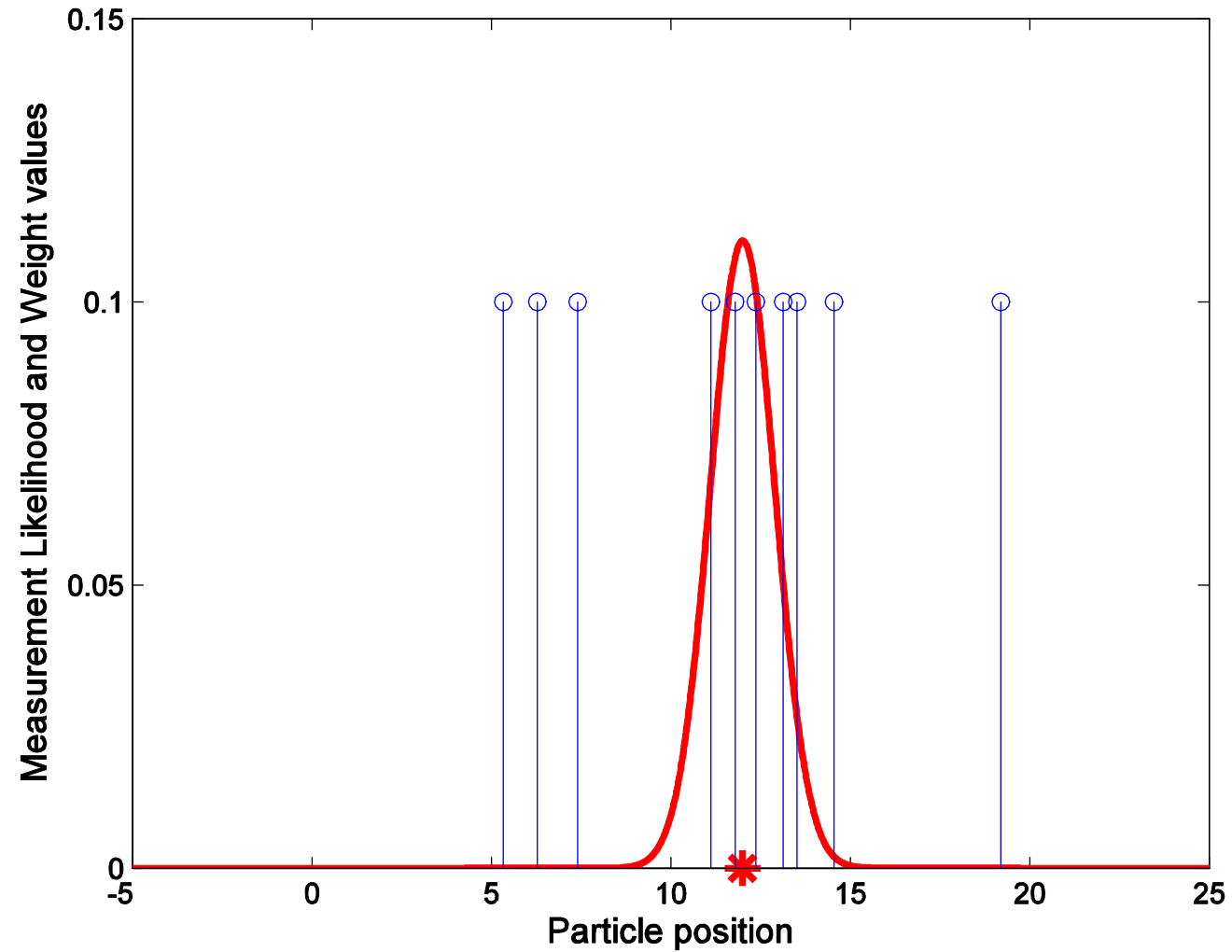
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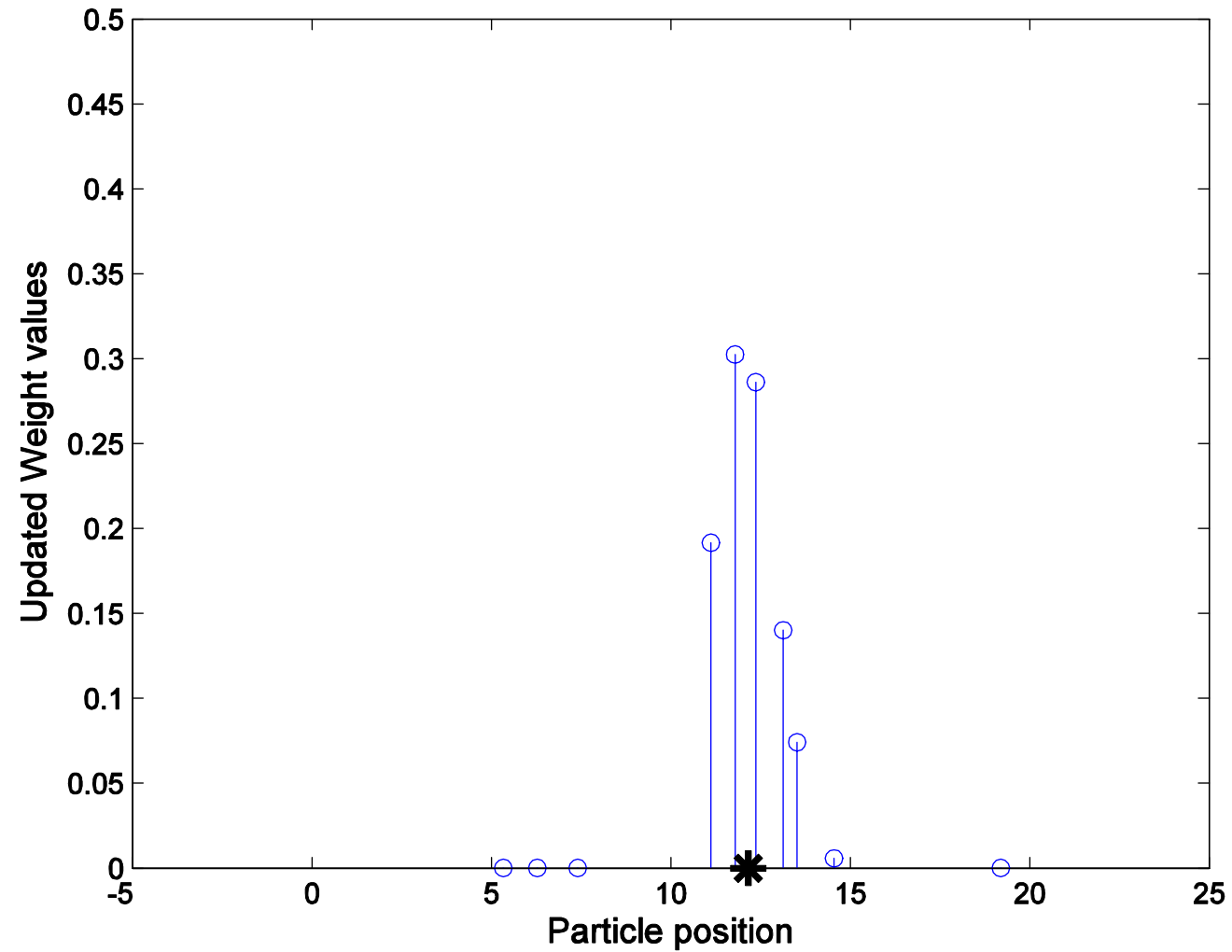
PARTICLE FILTER



PARTICLE FILTER



PARTICLE FILTER





STATE TRANSITION – DYNAMICS $p(x_{t+1}|x_t)$

- Vehicle dynamics with inertial signals (accelerometers and rate gyros) as inputs in a state space form

$$\begin{aligned}p_{t+1} &= p_t + T_s v_t + \frac{T_s^2}{2} R(q_t)^T (a_t + w_t^a) \\v_{t+1} &= v_t + T_s R(q_t)^T (a_t + w_t^a) \\q_{t+1} &= \expm\left(\frac{T_s}{2} S(\omega_t + w_t^\omega)\right) q_t\end{aligned}$$

where

a_t is the acceleration measurement (input),

ω_t is the gyro measurement (input),

$w_t^a \sim \mathcal{N}(0, R^a)$ and $w_t^\omega \sim \mathcal{N}(0, R^\omega)$ are sensor noises,

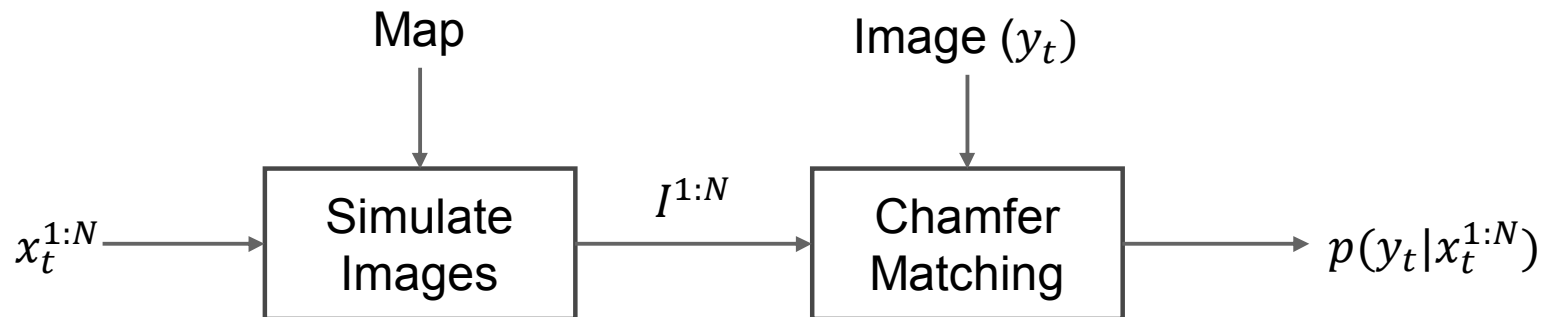
$R(q_t)$ encodes rotation between body and navigation frames,

$S(\cdot)$ is the mapping between a vector and a skew-symmetric matrix

LIKELIHOOD FUNCTION – MEASUREMENTS

$$p(y_t|x_t)$$

- The difficult part due to its construction
- Based on image matching with Chamfer method
- No analytical gradient – EKF is hard(er) to implement



LIKELIHOOD FUNCTION – MEASUREMENTS

- A simple example image from a camera mounted on the helicopter UAV
- Map image taken from the internet



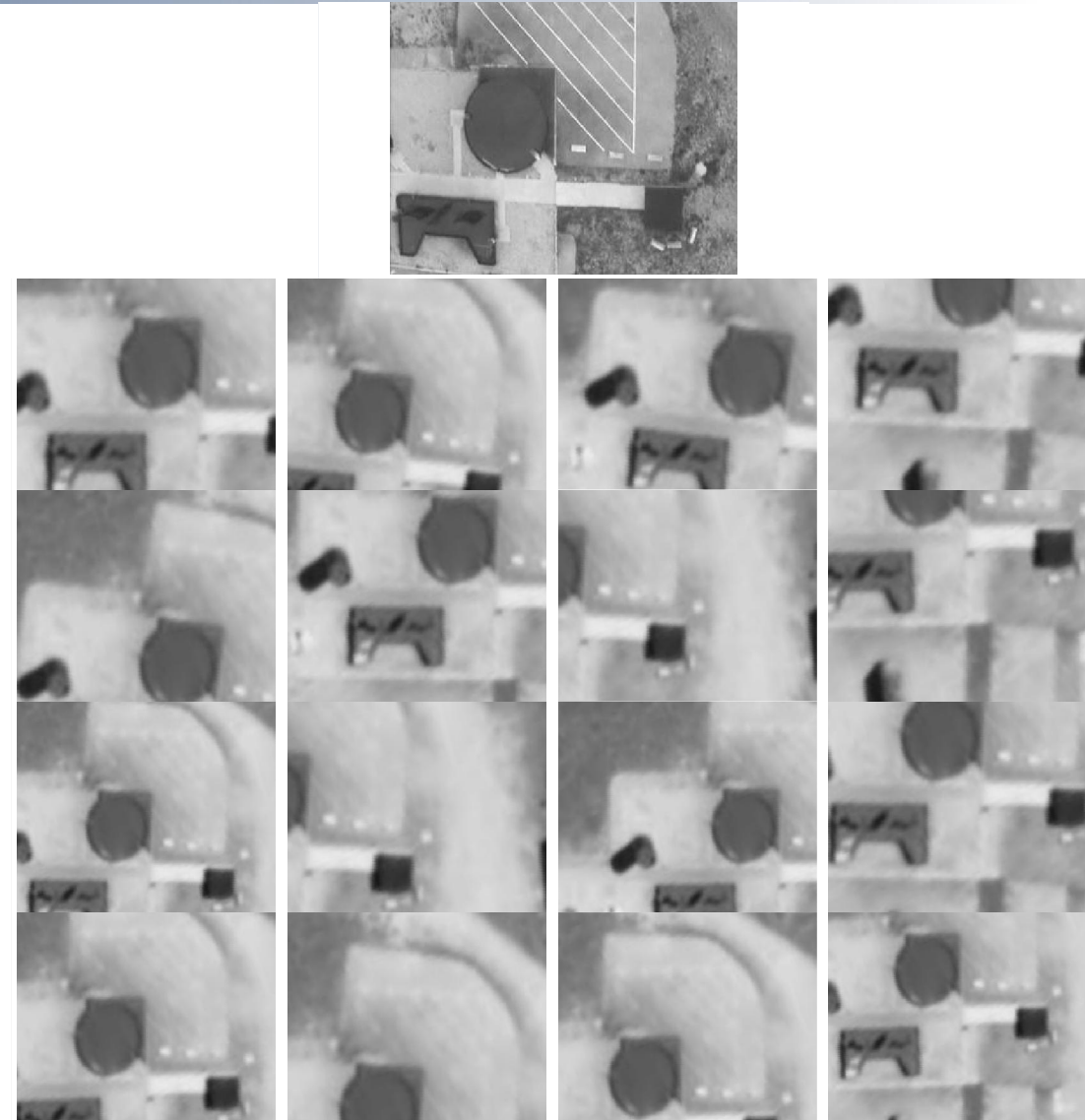
LIKELIHOOD FUNCTION – MEASUREMENTS

- A simple example image from a camera mounted on the helicopter UAV
- Map image taken from the internet
- Map is used to create simulated images given the position and heading hypothesis (represented with N particles, $i = 1:N$)

$$m_C^j = R(q_t^i) \left(p_t^i - m^j(1:3) \right)$$

$$\begin{bmatrix} u^j \\ v^j \end{bmatrix} = \mathcal{P}_C \left(\begin{bmatrix} m_C^j(1) \\ m_C^j(3) \\ m_C^j(2) \\ m_C^j(3) \end{bmatrix} \right)$$

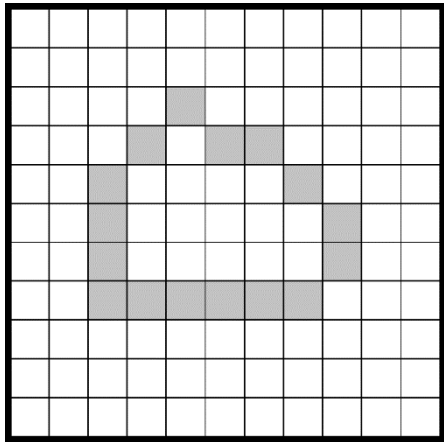
$$I^i(u^j, v^j) = m^j(4)$$



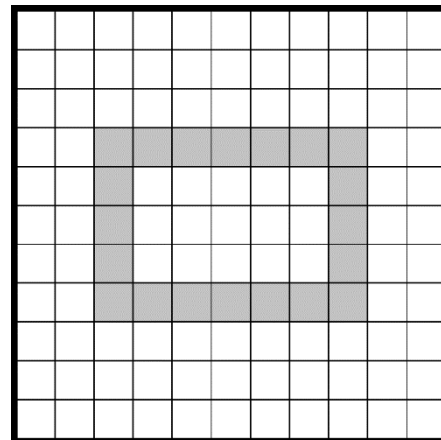
CHAMFER MATCHING

- A robust image matching method
- Based on the edge images and distance transform

Pattern



Template





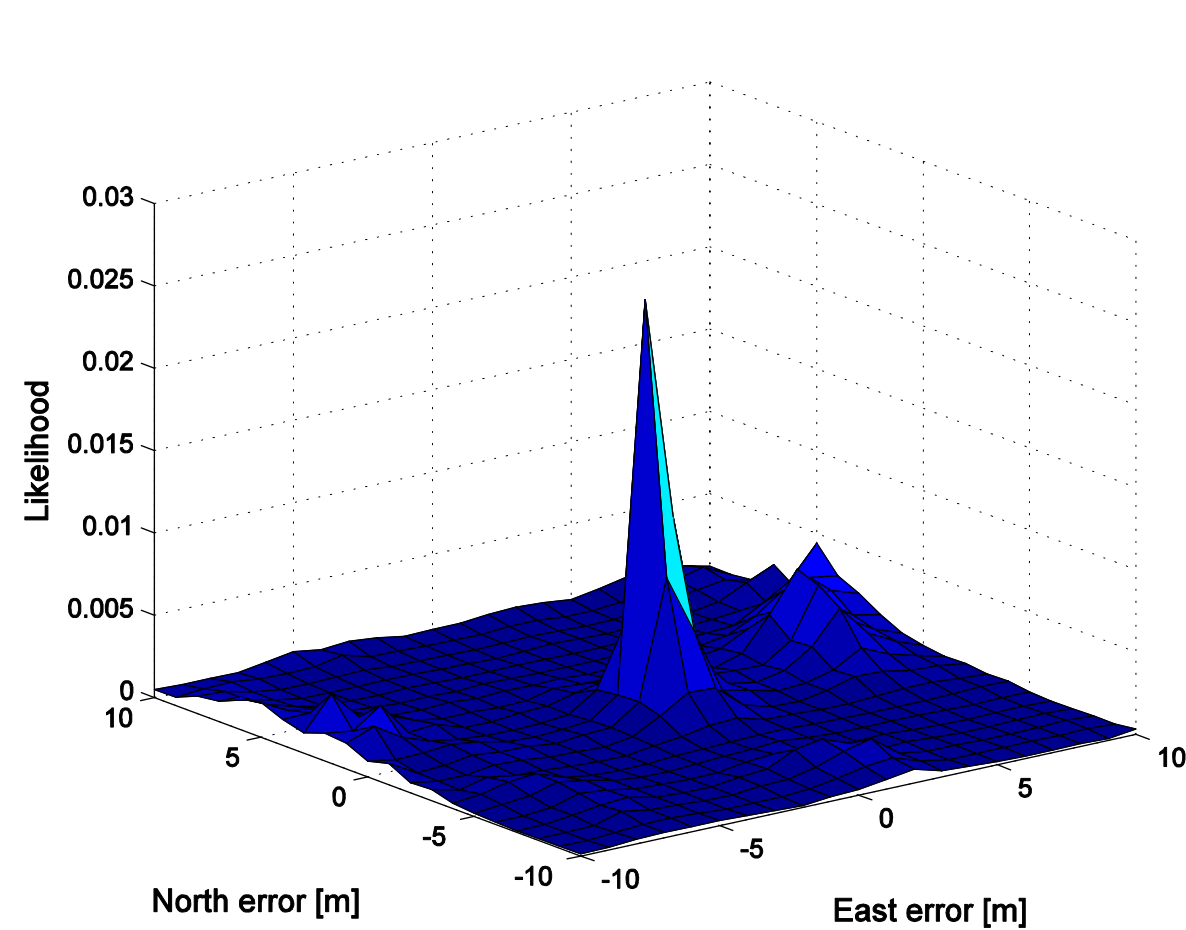
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CHAMFER MATCHING

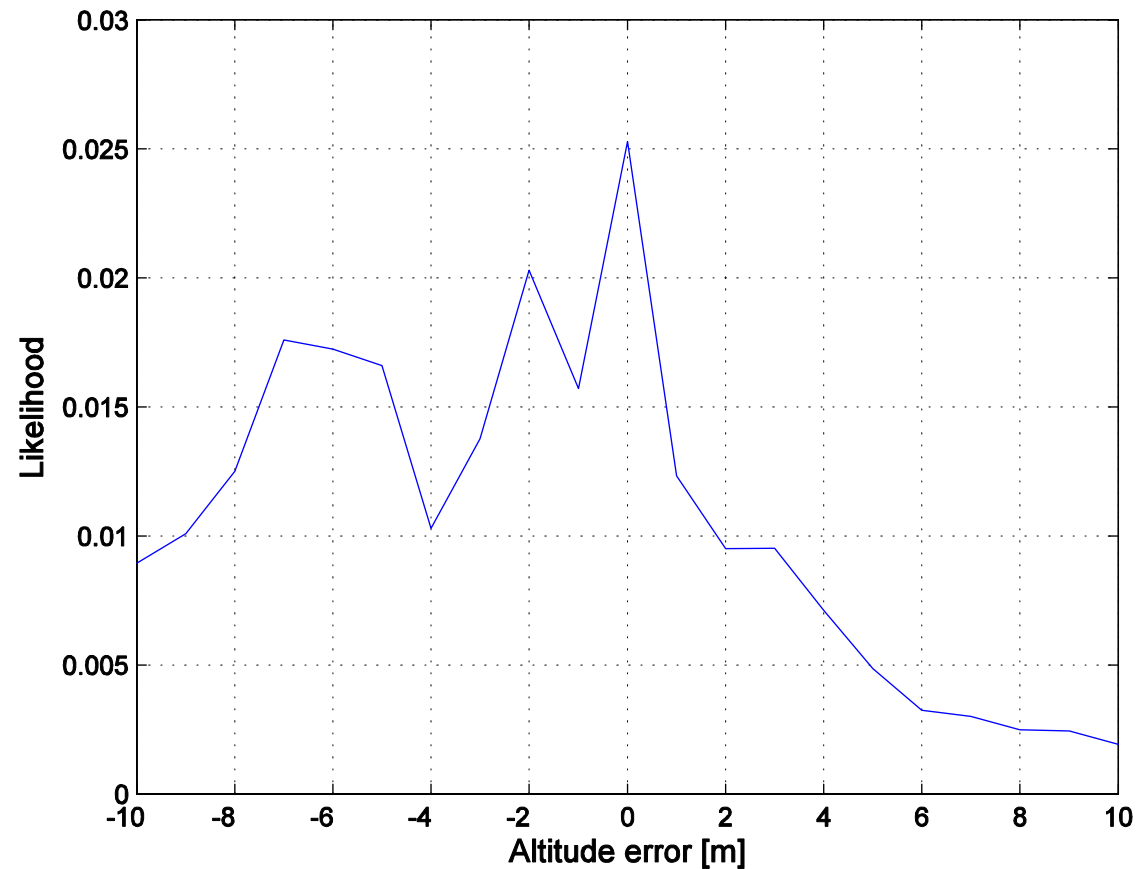
EVALUATION OF THE LIKELIHOOD – GOOD CASE



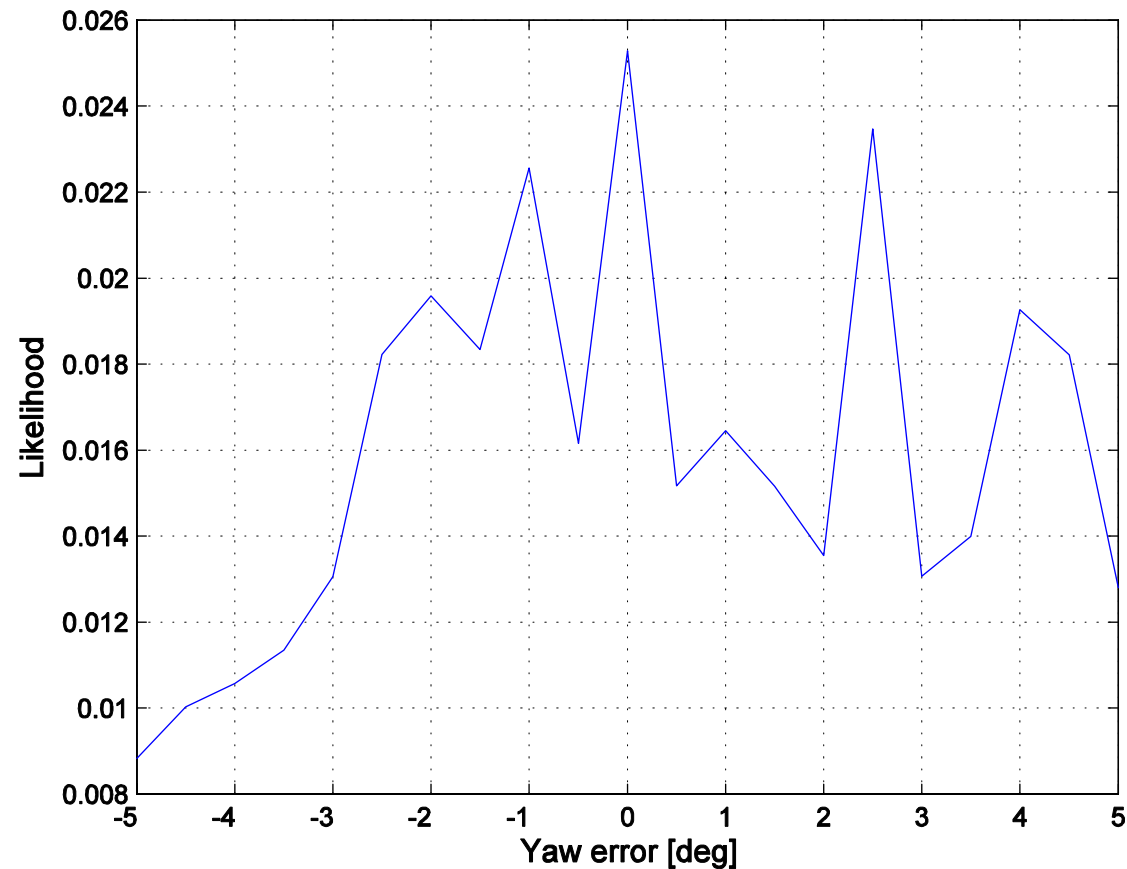
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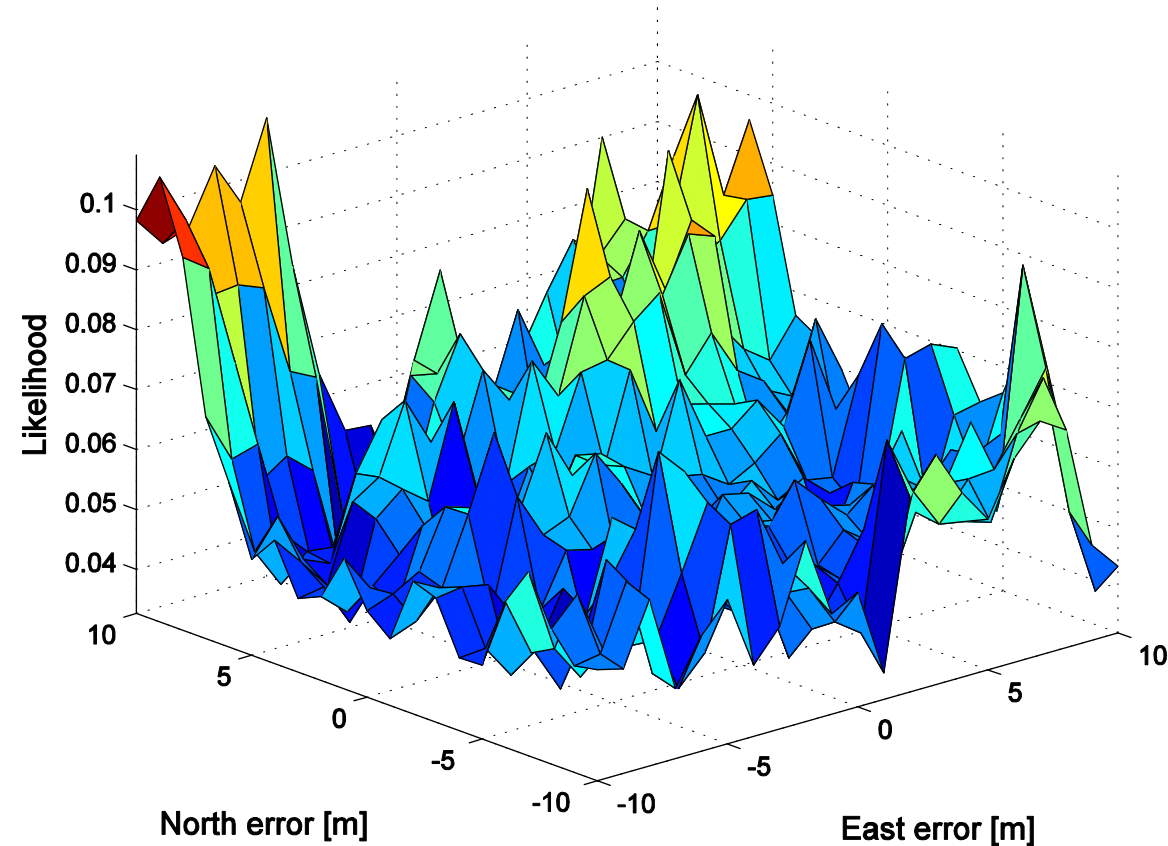
EVALUATION OF THE LIKELIHOOD – GOOD CASE



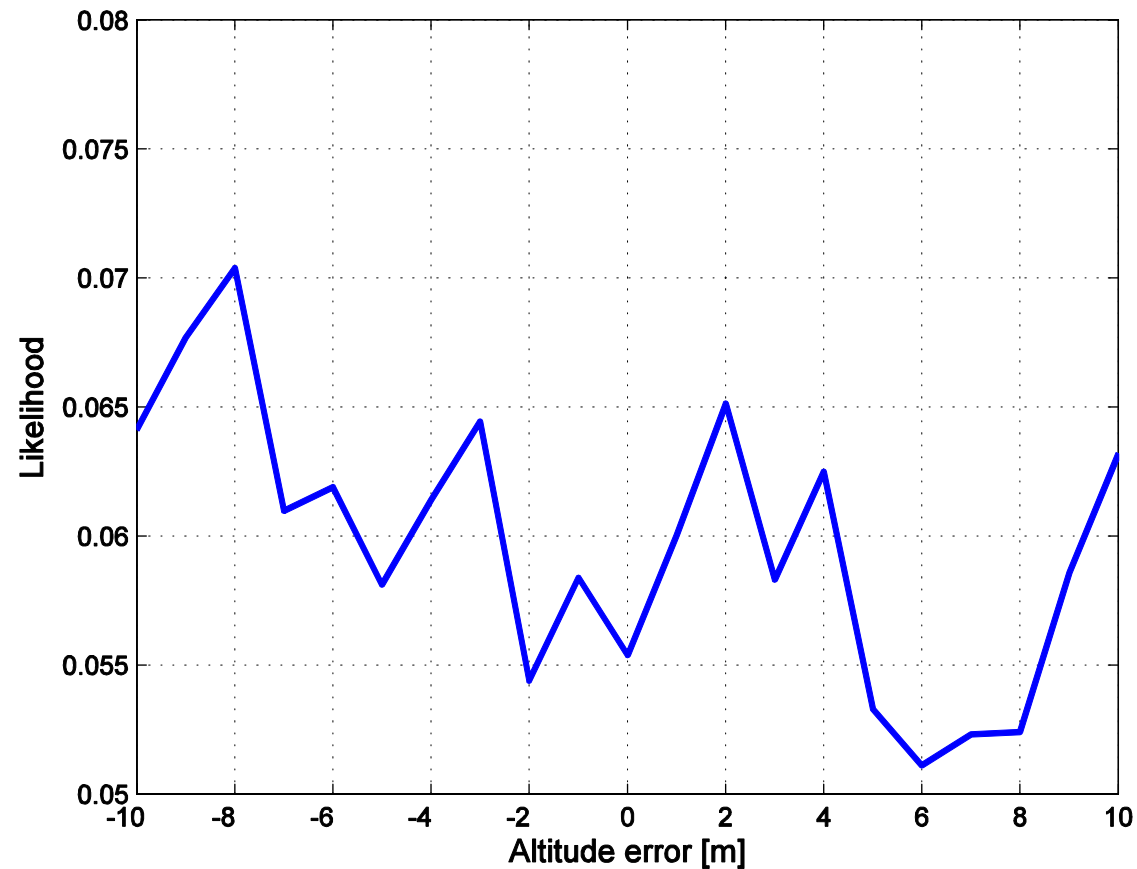
EVALUATION OF THE LIKELIHOOD – BAD CASE



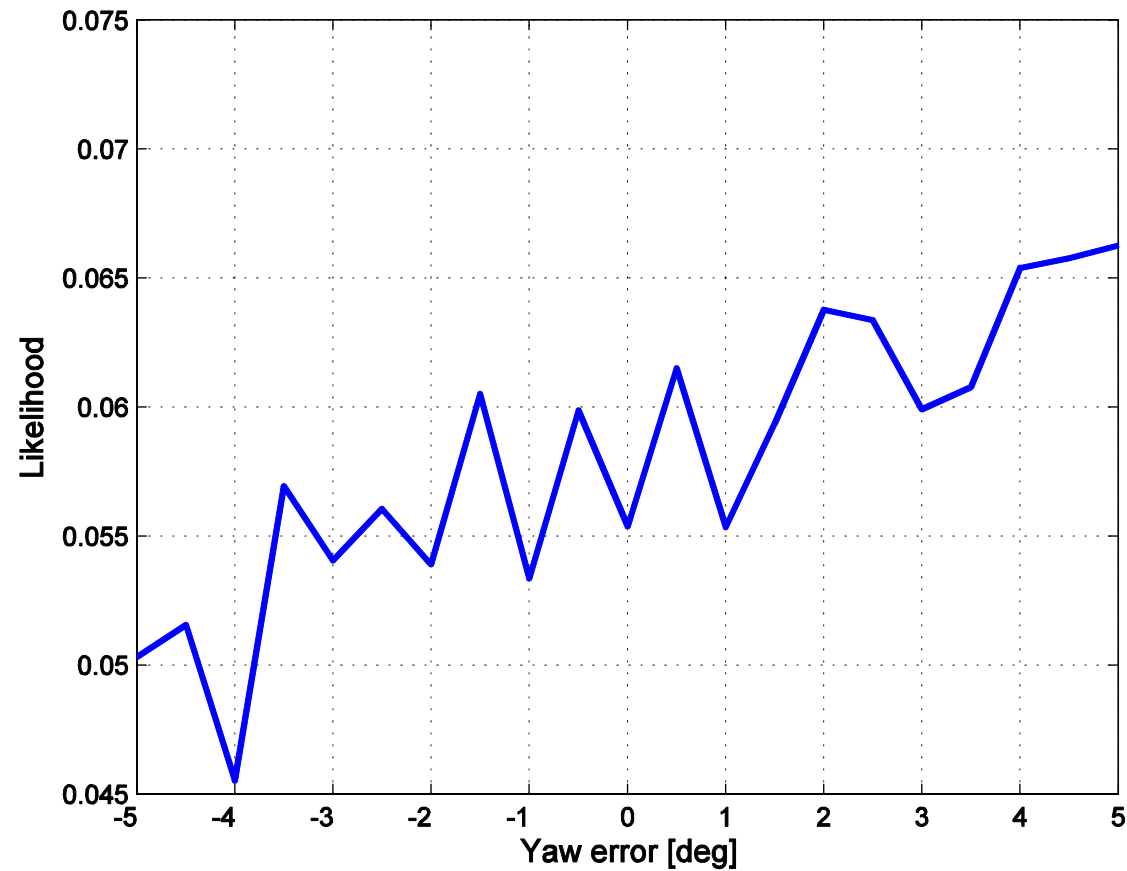
EVALUATION OF THE LIKELIHOOD – BAD CASE



EVALUATION OF THE LIKELIHOOD – BAD CASE



EVALUATION OF THE LIKELIHOOD – BAD CASE



CONCLUSIONS AND FUTURE WORK

- A method for estimation of an aerial vehicle's 3D-position and heading without GNSS is presented
- Evaluation concentrated on likelihood function
- Results are promising for now – informative images(!)
- Sensitivity to pitch and roll deviation from zero shall be studied
- Evaluation of the whole filter on real UAV data



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