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Static failure in components made of Al-Li AA2050 and Al AA7050 alloys

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Background and main issues

- AI-Li AA2050-T84 is superior to AI AA7050-T7451 regarding
 - Tensile strength
 - Stiffness
 - Density
 - Fatigure crack growth rate
- Cracks in AA2050 have shown a tendency to turn towards the material L-direction in
 - residual strength test
 - during fatigue crack growth
- Does the same phenomenon occur in static failure without previous crack growth?
- Are traditional sizing methods yet conservative?



Crack turning in residual fracture

Crack turning during fatigue

crack growth

Fracture

Fracture originating from notch without defect.



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Physical tests and numerical analyses

• Several non-standard tests were carried out in this investigation



- Physical tests
 - Reference tensile tests
 - Notched tensile tests
 - Bending tests
 - Bend-shear tests
- AI-Li AA2050-T84 and AI AA7050-T7451
- Test results were compared with
 - Analytical predictions
 - Finite element analyses









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Notched tensile tests

- Open and filled hole
- Material directions L, LT, ST
- Variety of D/W and thickness
- Failure load close to $F = R_m A_{net}$
- Tension and compression







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Notched bending tests

- A four point bending test rig was developed
- Specimens with geometric features
 - Beams with holes (bending)
 - Beams with notches (bending)
 - I-sections with holes (shear)
- Specimens made of
 - AA7050-T7451
 - AA2050-T84





Bending tests with holes

Bending tests with notches

Bending shear test with holes

1A4

X

AA7050-T7451

 Cracking along L in both materials

Notch sensitivity

• AA7050-T7451 went through a larger overall plastic deformation than AA2050-T85

Shear with stress concentration

AA2050-T84

IAI

displacement [mm]

1AI

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ST

Shear test with holes

- Material AA2050-T84
- Offset hole
- Primary crack through hole
- Secondary crack in homogeneously strained area

cross head displacement [mm]

Shear test – finite element simulation

Strength anisotropy:

- Example: $R_{m,ST} \neq R_{m,L}$ Relation between shear and normal strength
- Al-Li : $\frac{\sigma}{\tau} \approx 2$

Normal stress

 σ_{ST}

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• von Mises : $\frac{\sigma}{\tau} = \sqrt{3}$

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Analytical calculations

Predictions of failure loads were made with classical analytical methods

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Conclusions

- Fractures in AA7050-T7451 are in general perpendicular to the major principle stress
- Fractures in AA2050-T84 are prone to start or turn towards the L-direction
- The usual static sizing methods can be used
- Strength anisotropy must be accounted for in the sizing

