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# Static failure in components made of Al-Li AA2050 and Al AA7050 alloys

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# Background and main issues

- AI-Li AA2050-T84 is superior to AI AA7050-T7451 regarding
  - Tensile strength
  - Stiffness
  - Density
  - Fatigure crack growth rate
- Cracks in AA2050 have shown a tendency to turn towards the material L-direction in
  - residual strength test
  - during fatigue crack growth
- Does the same phenomenon occur in static failure without previous crack growth?
- Are traditional sizing methods yet conservative?



Crack turning in residual fracture

Crack turning during fatigue

crack growth

Fracture

Fracture originating from notch without defect.



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# Physical tests and numerical analyses

• Several non-standard tests were carried out in this investigation



- Physical tests
  - Reference tensile tests
  - Notched tensile tests
  - Bending tests
  - Bend-shear tests
- AI-Li AA2050-T84 and AI AA7050-T7451
- Test results were compared with
  - Analytical predictions
  - Finite element analyses









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#### Notched tensile tests

- Open and filled hole
- Material directions L, LT, ST
- Variety of D/W and thickness
- Failure load close to  $F = R_m A_{net}$
- Tension and compression







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#### Notched bending tests

- A four point bending test rig was developed
- Specimens with geometric features
  - Beams with holes (bending)
  - Beams with notches (bending)
  - I-sections with holes (shear)
- Specimens made of
  - AA7050-T7451
  - AA2050-T84









### Bending tests with holes





#### Bending tests with notches





# Bending shear test with holes

1A4

X

AA7050-T7451

 Cracking along L in both materials

Notch sensitivity

• AA7050-T7451 went through a larger overall plastic deformation than AA2050-T85

Shear with stress concentration

AA2050-T84

IAI

displacement [mm]

1AI





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**ST** 

### Shear test with holes



- Material AA2050-T84
- Offset hole
- Primary crack through hole
- Secondary crack in homogeneously strained area



cross head displacement [mm]



#### Shear test – finite element simulation

Strength anisotropy:

- Example:  $R_{m,ST} \neq R_{m,L}$ Relation between shear and normal strength
- Al-Li :  $\frac{\sigma}{\tau} \approx 2$

Normal stress

 $\sigma_{ST}$ 

11

• von Mises :  $\frac{\sigma}{\tau} = \sqrt{3}$ 





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# Analytical calculations

Predictions of failure loads were made with classical analytical methods





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#### Conclusions

- Fractures in AA7050-T7451 are in general perpendicular to the major principle stress
- Fractures in AA2050-T84 are prone to start or turn towards the L-direction
- The usual static sizing methods can be used
- Strength anisotropy must be accounted for in the sizing



